

Adjoint Sensitivities and Calibration of Hypersonic Flow Problems in SPARC

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SPARC: Sandia Parallel Aerodynamics and Reentry Code (T. Fischer, et al)

Goal: Create a credible full-system virtual flight-testing platform for hypersonic vehicles

Modeling

- Perfect and non-equilibrium thermal and chemical gas models
- Euler, Laminar, RANS, Hybrid RANS/LES, LES, and DNS
- Structured and Unstructured Finite Volume methods
- R&D in structured and unstructured high-order methods
- Simulate coupled ablation
- Couples to SIERRA for full-system thermal and structural analyses

Performance and Portability

- Performance Portability through Kokkos
- Good performance on x86, Arm, and GPU platforms
- Uses performance portable/scalable linear solvers from Trilinos
- Uses embedded geometry and inline mesh refinement

Credibility

- Validation with UQ against wind tunnel and flight test data
- Visibility and peer review by external hypersonics community





Background

Double-cone geometry



- Uncertainties in inflow (boundary) conditions required calibrating them against subset of data
- Problems are steady-state, but complex shock interactions requires use of time integration methods to find solutions (pseudo-transient)
- Due to simulation expense, calibration was conducted using surrogate models trained from samples of SPARC simulations
- ATDM/DPC project investigated tools for "embedded analysis"
 - Embedded sensitivity analysis via Sacado
 - Embedded derivative-based optimization/calibration using ROL
- FY22 ATDM Milestone:
 - Investigate embedded workflows for formulating and solving these calibration problems
 - Investigate the feasibility of solving these problems based on adjoint sensitivities to provide a foundation for distinguishing future capabilities
 - Leverage Sacado automatic differentiation (AD), Tempus time integration, and ROL optimization components







Double Cone calibrations

- Several experimental data sets used for prior validation of SPARC
 - Run35 (easy) CUBRC LENS-I shock tunnel, perfect gas, vibration & reaction equilibrium, Mach 11
 - Case 1 (moderate) CUBRC LENS-XX expansion tunnel, real gas, vibrational non-equilibrium, reaction equilibrium, Mach 12
 - Case 4 (hard)– CUBRC LENS-XX expansion tunnel, real gas, vibration & reaction non-equilibrium, Mach 12
- Physics: laminar flow, some dissociation, separation & reattachment, shock interactions
- Measurements: pressure and heat-flux on surface of the cone at several locations
 - Due to model form error, only use probe locations ahead of the separation region
- Prior inference with Bayesian methods and surrogate models complete and published (DOI:<u>10.2514/1.J059033</u>)







Embedded Optimization with ROL

- ROL Rapid Optimization Library (D. Ridzal et al)
 - Derivative-based optimization library in Trilinos focusing on PDE-constrained optimization

```
\min_{y} g(u, y) = 0
s.t. f(u, y) = 0
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 Here we use the trust region-CG optimization method using BFGS approximation to the Hessian and reduced-space formulation:

 $\min_{y} h(y), \quad h(y) = g(u(y), y) \quad \text{s.t.} \quad f(u(y), y) = 0$ $\frac{\partial h}{\partial y} = \frac{\partial g}{\partial u} \frac{\partial u}{\partial y} + \frac{\partial g}{\partial y} \quad \text{s.t.} \quad \frac{\partial f}{\partial u} \frac{\partial u}{\partial y} + \frac{\partial f}{\partial y} = 0$

Double cone calibration objective function:

$$g(u, y) = \sum_{i=1}^{N_q} \left[s_q(q(x_i; u, y) - \tilde{q}_i) \right]^2 + \sum_{i=1}^{N_p} \left[s_p(p(x_i; u, y) - \tilde{p}_i) \right]^2 + \left[s_h(h_0(y) - \tilde{h}_0) \right]^2 + \left[s_P(P_{Pitot}(y) - \tilde{P}_{Pitot}) \right]^2$$

- Calibration parameters: inflow boundary conditions for density, velocity
 - Determined objective function is insensitive to temperature through local sensitivity analysis



Pseudo-transient Nonlinear Solvers

SPARC employs *pseudo-transient* nonlinear solvers to compute steady-state flows:

$$M\dot{u} + f(u, y) = 0 \implies \frac{1}{\Delta t_k}(Mu_{k+1} - Mu_k) + f(u_{k+1}, y) = 0$$
 (BDF1)

Called pseudo-transient because time-step residual is not driven to zero each step
Typically small number of Newton iterations per time step, using an approximate Jacobian:

$$\begin{split} \tilde{f}_{k}(u_{k+1},y) &\equiv \frac{1}{\Delta t_{k}}(Mu_{k+1} - Mu_{k}) + f(u_{k+1},y), \\ \left(\frac{1}{\Delta t_{k}}M + J(u_{k+1}',y)\right) \Delta u_{k+1}' &= -\tilde{f}_{k}(u_{k+1}',y), \quad u_{k+1}'' = u_{k+1}' + \Delta u_{k+1}', \\ J(u_{k+1}',y) &\approx \frac{\partial f}{\partial u}(u_{k+1}',y) \end{split}$$

- Each linear system approximately solved using a light-weight linear solver (typically block tridiagonal solver provided by lfpack2)
- Manually specified sequence of increasing time-step sizes (called a run-schedule)
 - Requires O(10,000-100,000) time steps



Pseudo-transient Sensitivities

Similar pseudo-transient approach possible for forward sensitivity equations

$$M\dot{Z} + \frac{\partial f}{\partial u}(u, y)Z + \frac{\partial f}{\partial y} = 0, \quad Z = \frac{\partial u}{\partial y} \implies \frac{1}{\Delta t_k}(MZ_{k+1} - MZ_k) + \frac{\partial f}{\partial u}(u_{k+1}, y)Z_{k+1} + \frac{\partial f}{\partial y}(u_{k+1}, y) = 0$$

Results in similar solution approach:

$$F_{k}(Z_{k+1}, u_{k+1}, y) \equiv \frac{1}{\Delta t_{k}}(MZ_{k+1} - MZ_{k}) + \frac{\partial f}{\partial u}(u_{k+1}, y)Z_{k+1} + \frac{\partial f}{\partial y}(u_{k+1}, y),$$
$$\left(\frac{1}{\Delta t_{k}}M + J(u_{k+1}', y)\right)\Delta Z_{k+1}' = -F_{k}(Z_{k+1}', u_{k+1}', y), \quad Z_{k+1}'' = Z_{k+1}' + \Delta Z_{k+1}',$$

- Combined approach: simultaneously solve state and sensitivity equations each time-step
- Allows use of same run-schedule for state and sensitivity equations
- Sensitivity approach provided by Tempus time integration package (C. Ober et al)
 - Requires users to implement needed partial derivatives
- Sensitivity residual computed using forward-mode Automatic Differentiation (AD) with Sacado package
 Use "tangent mode" to compute (^{∂f}/_{∂u}) Z + ^{∂f}/_{∂v} without explicitly forming ^{∂f}/_{∂u}



Double Cone Sensitivity Analysis

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- Sensitivity of objective function with respect to inflow density, velocity, and temperature for Case 1
 - Shows objective function is not sensitive to temperature and can be removed from calibration set



Adjoint Sensitivities

- Recall ROL requires computing the reduced-gradient $\frac{\partial h}{\partial y} = \frac{\partial g}{\partial u} \frac{\partial u}{\partial y} + \frac{\partial g}{\partial y}$ s.t. $\frac{\partial f}{\partial u} \frac{\partial u}{\partial y} + \frac{\partial f}{\partial y} = 0$
- Adjoint sensitivity approach: $\frac{\partial h}{\partial y} = \frac{\partial g}{\partial u} \left(\frac{\partial f}{\partial u}\right)^{-1} \frac{\partial f}{\partial y} + \frac{\partial g}{\partial y} \implies \left(\frac{\partial h}{\partial y}\right)^T = \left(\frac{\partial f}{\partial y}\right)^T \left(\frac{\partial f}{\partial u}\right)^{-T} \left(\frac{\partial g}{\partial u}\right)^T + \left(\frac{\partial g}{\partial y}\right)^T$
- Requires solving linear system of the form: $\left(\frac{\partial f}{\partial u}(u_{\infty}, y)\right)^T w = \left(\frac{\partial g}{\partial u}(u_{\infty}, y)\right)^T$
- While a pseudo-transient approach similar to forward sensitivities is possible, found it was not effective on these problems
 Full transient adjoint and pseudo-transient adjoint capabilities provided by Tempus
- Found a Newton-GMRES approach to be the most effective
 - Apply Newton's method to linear system and solve linear system at each step using GMRES (provided by Belos)
 - Precondition GMRES using SPARC's block tri-diagonal solver applied to the native (approximate) Jacobian-transpose
 - Because of ill-conditioning, multiple Newton iterations (O(10)) are required to achieve small linear residuals with a bounded number of linear iterations per solve (O(100)) (equivalent to iterative refinement and restarted GMRES)
 - Implemented through Tempus interface to SPARC using a single time-step, not including transient terms
- Need to compute analytic adjoint matrix $\left(\frac{\partial f}{\partial u}\right)^T$
 - Leverage existing Sacado tangent capabilities to compute $\frac{\partial f}{\partial u}V$ for any matrix V
 - Use graph coloring provided by Zoltan2 to find *V* with a small number of columns
 - Form adjoint through explicit transpose



Run 35 Calibration (Fine Mesh)

	Experiment	ROL Converged	Bayesian (90% Cl)
Density	0.5848	0.589	0.574
[g/m^3]	(0.5439, 0.6257)		(0.5471, 0.6209)
Velocity	2545	2506	2490
[m/s]	(2469, 2621)		(2441, 2653)

Run 35 data set



- Perfect gas
- Vibration equilibrium
- Reaction equilibrium



Heat flux comparison



Heat flux comparison





Case 1 Calibration (Medium Mesh)

	Experiment	ROL Converged	Bayesian (90% Cl)
Density	0.4990	0.433	0.4897
[g/m^3]	(0.4641, 0.5339)		(0.4328, 0.5645)
Velocity	3246	3540	3340
[m/s]	(3149, 3343)		(3211, 3654)

Case 1 data set

- Moderate problem
- Real gas
- Vibration non-equilibrium
- Reaction equilibrium





Pressure comparison





Case 4 Calibration (Coarse Mesh)

	Experiment	ROL Converged	Bayesian (90% Cl)
Density	0.9840	0.866	0.8608
[g/m^3]	(0.9151, 1.053)		(0.7996, 1.0396)
Velocity	6479	6940	7060
[m/s]	(6285, 6673)		(6380, 7089)

Case 4 data set

- Hardest problem
- Real gas
- Vibration non-equilibrium
- Reaction non-equilibrium



Heat flux comparison



0

0.00

0.05

0.10

x [m]

0.15

0.20



Computational Cost Comparisons

Run-time (in seconds) for forward (FSA) and adjoint (ASA) sensitivity computations.

Memory high-water mark (in MB) for forward and adjoint sensitivity computations.

Problem	Mesh	State (SPARC)	State (Tempus)	Adjoint	ASA Total	FSA Total	x-Speedup
Run 35	Coarse	58	79	3	82	208	2.5
	Medium	110	149	3	152	474	3.1
	Fine	622	829	13	842	2311	2.7
Case 1	Coarse	440	527	6	533	1890	3.5
	Medium	1709	2146	12	2158	7393	3.4
	Fine	12440	15219	41	15259	51628	3.4
Case 4	Coarse	558	727	7	734	2861	3.9
	Medium	7139	9081	17	9099	33121	3.6
	Fine	15024	18976	44	19020	68221	3.6

Tempus-based state integration about 25% slower than SPARC-native implementation

- Due to extra residual computations for consistent computation of CFL-limited time step required by Tempus
- Cost of the adjoint solve is insignificant compared to forward state integration
- Adjoint sensitivity is about 3 times faster than forward approach on these problems
 - Directly translates into comparable speedup for ROL calibration
- Adjoint approach requires substantially more memory
 - Due to cost of storing true adjoint matrix and Tempus implementation requiring several copies of this matrix
 - A limiting factor for small memory environments such as GPUs



ROL and Surrogate-based Inversion Comparisons

- The ROL-based calibration is a deterministic inversion approach which doesn't directly provide estimates of uncertainty
- Assuming
 - Qols (heat flux, pressure, total enthalpy, Pitot pressure) differ from experimental values by additive Gaussian noise
 - This noise is sufficiently small such that QoIs depend approximately linearly on the calibrated parameters
- Then
 - The posterior of the calibrated parameters is (approximately) Gaussian
 - Solution of the ROL calibration problem is equivalent to MLE for the mean of the posterior
 - The inverse of the Hessian of the objective function (negative log likelihood) at the calibrated parameters is an estimate of the covariance of the posterior
- Does this provide a useful estimate of uncertainty?
- Estimate Hessian by differencing gradient after termination of ROL calibration (3 extra gradient computations)



Run 35



MCMC Gaussian Expt MAP Opt	\wedge	
2.0 2.2 2.4	2.0 2.8 3.0 3.	2

2	un	35	data	set
-	U	~ ~	0.0.00	~ ~

- **Easiest problem**
- Perfect gas
- Vibration equilibrium
- Reaction equilibrium

Summary	$ ho_{\infty}/$	ρ_{norm}	U_{∞} /	U_{∞}/U_{norm}		
	MCMC	Deterministic	MCMC	Deterministic		
MAP	5.74	5.89	2.490	2.506		
Mean	5.83	5.89	2.548	2.506		
Median	5.82	5.89	2.55	2.506		
IQR	(5.62, 6.03)	(4.24, 7.54)	(2.5, 2.6)	(2.32, 2.69)		
90% CI	(5.47, 6.21)	(1.88, 9.9)	(2.44, 2.65)	(2.06, 2.95)		

- Deterministic calibrated parameters
 - Agree well with MCMC Map/Mean/Median
 - Are within the range of experimental uncertainty
- IQR for velocity comparable, but 90% CI is too wide
- Both IQR and 90% CI for density are way too wide

Case 1



e	
- 52	- MAP
2.0	Opt Expt
1.5	
1.0	
0.5	
0.0	
	2.0 2.5 3.0 3.5 4.0 4.5 5.
	11/11

Case 1 data set

- Moderate problem
- Real gas
- Vibration non-equilibrium
- Reaction equilibrium

Summary	$ ho_{\infty}/$	ρ_{norm}	U_{∞}/U_{norm}		
	MCMC	Deterministic	MCMC	Deterministic	
MAP	4.897	4.33	3.34	3.54	
Mean	4.96	4.33	3.44	3.54	
Median	4.95	4.33	3.45	3.54	
IQR	(4.6, 5.31)	(2.86, 5.79)	(3.36, 3.53)	(3.28, 3.79)	
90% CI	(4.33, 5.64)	(0.73, 7.96)	(3.21, 3.65)	(2.92, 4.16)	

- Deterministic calibrated parameters
 - Agree somewhat with MCMC Map/Mean/Median
 - Are outside the range of experimental uncertainty
- IQR for velocity somewhat comparable, but 90% CI is too wide
- Both IQR and 90% CI for density are way too wide

Case 4



5	-	мсма		Á			
1.5	=	Gauss MAP Opt Expt	sian				
2 -							
- 6			X				
0.0	-	-	1	l	-	-	_
	4	5	6	7	8	9	10
	4	5	0	í	0	3	

Case 4 data set

- Hardest problem
- Real gas
- Vibration non-equilibrium
- Reaction non-equilibrium

Summary	$ ho_{\infty}/$	ρ_{norm}	U_{∞}/U_{norm}		
	MCMC	Deterministic	MCMC	Deterministic	
MAP	8.608	8.66	7.06	6.94	
Mean	9.186	8.66	6.8	6.94	
Median	9.169	8.66	6.834	6.94	
IQR	(8.57, 9.81)	(6.32, 11.0)	(6.67, 6.96)	(6.52, 7.36)	
90% CI	(8.0, 10.4)	(2.98, 14.37)	(6.38, 7.09)	(5.92, 7.95))	

- Deterministic calibrated parameters
 - Agree somewhat with MCMC Map/Mean/Median
 - Are outside the range of experimental uncertainty
- IQR for velocity somewhat comparable, but 90% CI is too wide
- Both IQR and 90% CI for density are way too wide

Conclusions

- Demonstrated adjoint sensitivities for hypersonic flows can be successful and facilitate embedded calibration/optimization
 - Capabilities are built-in to SPARC and don't require construction of surrogate models
 - Provides similar calibrated parameters to surrogate/Bayesian approach (uncertainty estimates not useful though)
 - Adjoint sensitivity calculation is essentially free compared to forward state solve
- Approach leverages multiple Trilinos capabilities
 - Sacado AD
 - ROL embedded optimization
 - Tempus time integration
 - Zoltan2 graph coloring
 - Ifpack2 block tri-diagonal solver
 - Belos GMRES
- Techniques provide a foundation for future distinguishing capabilities
 - Field inversion
 - Model form error estimation
 - Construction of ML-based turbulence model closures
 - Shape optimization



Backup Slides



Sacado: AD Tools for C++ Applications

Automatic differentiation package in Trilinos (Phipps and Gay)

Operator overloading-based approach

- Sacado provides C++ data types implementing AD
- Type of variables in code replaced by AD data type
- AD object for each variable stores value of that variable and its derivatives
- Mathematical operations replaced by overloaded versions implementing chain-rule
- Expression templates reduce overhead

Primary tools are Sacado's forward mode (a.k.a. tangent mode) AD tools

- Integrates with Kokkos for efficient differentiation of thread-parallel programs
- Compute sparse Jacobian's for finite element-type codes by differentiating at element level and manually assembling global Jacobian
- Global Jacobian vector products





Iso-velocity adjoint surface for fluid flow in a 3D steady MHD generator in Drekar computed via Sacado (Courtesy of T. Wildey)

Adjoint Flow Visualizations (Run35, Medium Mesh)



Adjoint Flow Visualizations (Run35, Medium Mesh)



Multi-start Optimization

- Case 1 ROL-based calibration results in calibrated parameters outside the range of experimental uncertainty
 - Whereas surrogate-based calibration was on the edge
- Is this due to the problem formulation, or are there possibly multiple local minima?
- Executed ROL inversions (on medium mesh) with 10 initial guesses to find out
- All initial guesses converge to same minimum
- As in Case 4, the freestream conditions quoted by experimentalists are likely wrong



Adjoint Sensitivity Accuracy Comparisons



Forward and adjoint sensitivity error on a coarse mesh blunt wedge regression test problem.



Comparison between adjoint sensitivities and finite differences on a coarse mesh blunt wedge regression test problem.



- Explored (combined) forward and adjoint sensitivity accuracy on several test problems
 - Coarse mesh blunt wedge regression test problem
 - Coarse mesh Run 35 problem
- Very similar accuracy between methods across wide range of solver tolerances.
- Sensitivities also agree with finite differences to expected precision, verifying correctness

